

PILOTS NOTES FIREFLY T67M-MkII

SECTION 3 NORMAL PROCEDURES

CONTENTS

3.1	Before	Starting the Engine	. 3-1
	3.1.1 3.1.2	Initial Check External Check	. 3-2
3.2	Starting t	the Engine	3-4
	3.2.1 3.2.2	Pre-start Cockpit Checks Starting the Engine and After Start Checks	. 3-4 . 3-5
3.3	Taxying		
3.4	Testing	the Engine	3-7
3.5		e off Vital Actions	
3.6		f and Climb	
3.7	Erect S	pin Recovery	2 10
	3.7.1 3.7.2 3.7.3	Standard Recovery Technique Non-Standard Recovery Aerobatics or Spinning - Gyro Instruments	3-10
3.8	Practice	Forced Landings	
3.9	Rejoin (Checks	3-11
3.10		Checks and Speeds	
	3.10.1 3.10.2 3.10.3	Downwind Checks Circuit Speeds Final Checks	3-11 3-12 3-12
3.11	After La	nding	
	3.11.1 3.11.2 3.11.3	Checks After Landing Stopping the Engine Fitting Flying Control Lock (Mod 435)	3-12

Last effective page is P.3-13

>



INTENTIONALLY BLANK

CAA Approved July 1993 All TP.T67M-MkII/FM



3.1 BEFORE STARTING THE ENGINE

3.1.1 Initial Check

Check loading and C of G (Section 7.1.1)

Approaching the aircraft: Chocks, towing arm, fire axe, fire extinguisher stowed, pitot cover, snow/ice/hoar frost, obstructions, aircraft attitude, obvious leaks.

Cockpit

Control lock Remove from aircraft
Parking Brake On
Magnetos Off, key out
Master switch On
Alternator warning Cancel flasher
Pitot heater On for 20 secs
Strobe light On - check - off
Landing lights On - check - off
Trim Note position
Stall warning Check light/horn
Pitot head Check heat
Pitot heater Off
(Night flying) Nav lights on - check, landing lights on - check both - off
(Structural temperature Press test switch in hot conditions) Check structural temperature - on OAT gauge below 50°C (For Post Mod 734B/D see Supplement F)
Master switch Off

CAUTION

Strobe position light not to be used in cloud or mist or on the ground.



3.1.2 External Check (ref. illustration 8.3 Principal Features) Start at left wing inboard trailing edge.

Left wing

Flap Condition, play, stiff nut

Undercarriage (rear) Tyre, torque link, brake leaks

Aileron Condition, movement, play, stiff nut,

drains

Wing Condition, drains

Wingtip Nav light

Leading edge Condition

Fuel cap Correctly fitted and locked

Fuel drain Check for water contamination

Access panel Security

Pitot head Remove cover/hole clear

Undercarriage (front) Condition/extension. Tyre creep/

inflation/condition. Brakes -

leaks/damage

Flap underside Condition, drains

Forward fuselage

Fresh air intake Clear

Cowling Pt side Security, 7 fasteners, 2 pins, oil leaks

Landing lights Undamaged

Propeller Condition, spinner

Nosewheel Condition, extension, tyre-cuts/creep/

inflation

Engine air inlet Check foam filter is clean

Cowling Stbd side Security, 6 fasteners, 2 pins

Oil Contents, panel secure

Fresh air intake Clear. Temp. probe



3.1.2 External Check (continued)

Right Wing

Leading edge Condition

Fuel cap Correctly fitted and locked

Fuel drain Check for water contamination

Undercarriage (front) Condition, extension. Tyre-creep/

inflation/condition. Brak

damage/leaks

Flap underside Condition, drains

Wing surfaces Condition

Access panel Security

Wingtip Nav light

Aileron Condition, movement, play, stiff nut,

drains

Wing Drains

Undercarriage (rear) Tyre, torque link, brake-leaks

Flap Condition, play, stiff nut

Nav aerials (if fitted) Secure/undamaged

Rear fuselage

Canopy stbd side Cracks, clean

Static vent starboard Plug out, clear

VHF aerial (if fitted) Secure/undamaged

Fin fairing Secure

Elevator Condition, movement, play, drains

Inspection cover Secure (side)

Strobe light Condition

Rudder DO NOT MOVE

Condition stiff nuts. Nav light

Trim tab Position, stiff nut, security, play

P.3-3 CAA Approved July 1993 A11 TP.T67M-MkII/FM



3.1.2 External Check (continued)

Rear fuselage (continued)

Tail bumper Unmarked

Static vent port Plug out, clear

Canopy port side Cracks, clean

3.2 STARTING THE ENGINE

3.2.1 Pre-start Cockpit Checks

Cockpit Check for loose articles
Rudder pedals Adjust for leg length
Harness Strap in (5 straps). If solo secure RH harness
Helmet/headset Plugged in
Controls (ail/elev) Full and free movement
Lights All off
Radios Off
Avionics Off
Fuel pump Off
Alternator Off
Master switch On
Intercomm On
Alternator warn Cancel
Pitot heat Off
Accelerometer Reset
Manifold pressure Note
Clark



PILOTS NOTES FIREFLY T67M-MkH

	3.2.1	Pre-start Cockpit Checks (Continue	d) _
		ASI	. Zero
		VSI	. ±100ft/min
>		Emergency static vent (Mod 485	. Closed
		Circuit breakers	. All in
		Throttle	. Check full movement leave closed
		Propeller	. Check full movement leave at max RPM
		Mixture	Check full movement leave at cut-off
		Fuel contents	Check (both gauges wing tank A/C)
•		Fuel cock	On (select tank with lowest quantity)
		Parking brake	On (Pump brakes)
		Flap	·
		Trim	Check full range and leave neutral
		Canopy	Secure
		Propeller	Clear
	3.2.2	Starting the Engine and After Start	Checks
	Engine	hot or cold	
		Mixture	Full rich
		Booster pump	On
		Throttle	Open (1/4 inch to 1/2 inch) until a slight fuel pressure is indicated on the fuel pressure gauge
		Booster pump	Off
		Mixture	Lean to cutoff
		Magneto	Left
		Starter	Press; (check starter warning light on during start); release when engine fires



PILOTS NOTES FIREFLY T67M-MkII

3.2.2 Starting the Engine and After Start Checks (continued)

Engine hot or cold (continued)

Mixture Slowly to full rich

Magneto Both

Starter warning Check out

RPM Set 1200 to warm up

Oil pressure Risen within 30 secs, if not,

magneto off

Fuel pressure Check

Magneto Check for dead cut

Alternator On

Radios

}..... As required

Suction Indicating

Horizon Erecting - adjust datum

DI Synchronise

Radio Check on 2 freqs if possible

Obtain taxy clearance

Altimeter Check setting/indications

Ammeter Shows positive charge

Alternator failure warning Check light out

Canopy Closed and locked

CAUTION

Should starter warning light fail to extinguish after starter button is released SHUT DOWN ENGINE and establish cause.

3.2.2 Starting the Engine and After Start Checks

AlternatorOn

<u>NOTE</u>

Avoid long periods of operation with the throttle at idle as this can lead to spark plug fouling.

,





3.4 TESTING THE ENGINE

CAUTION

WHEN CLOSING THE CANOPY PRIOR TO FLIGHT, CHECK ALIGNMENT OF WITNESS LINE ON CANOPY OPEN/CLOSE PLACARD AND BOTTOM OF RELEASE HANDLE, TO ENSURE THAT THE LATCH MECHANISM IS IN THE FULLY LOCKED POSITION

Canopy Closed and locked
Parking brake On (Pump brakes)
Safety Clear behind
Fuel cock Check on (Change tanks)
Fuel pressure 0.5 to 8 psi
Oil pressure Green 4.2 to 6.2 bar
Oil temp Green 40°C to 118°C
Cylinder head temp Green 100°C to 230°C
RPM Set 1800 RPM
Suction Green (4.5 to 5.5 in Hg)
Magneto drop Max 175 RPM, no more than 50 RPM difference between L and R
Propeller Exercise pitch control 4 times RPM drop not more than 500
Idling Check idling 800 RPM minimum

NOTE

Oil Pressure During Normal Operation Because of the greater length of the oil flow path from the sump to the oil pump, the pump has to work harder than normal to draw oil through these lines: the resultant pressure drop through these lines results in a lowered oil pressure. This effect will be more marked when the engine is cold and unlike a standard engine, the indicated oil pressure will normally tend to rise as the engine warms up. Thus it is not necessarily an indication of trouble if the oil pressure minima are only just met on start-up.



3.3 TAXYING

Check brakes immediately.

Check full rudder travel whilst taxying.

Check compasses and horizon and turn and slip for correct indications during turns.

3.4 TESTING THE ENGINE

Parking brake On
Safety Clear behind - Canopy locked
Fuel contents Check (Both gauges)
Fuel cock Check on (Change tanks)
Fuel pressure Indicating
Oil pressure Green (4.2 to 6.2 bar)
Oil temp Green (40°C to 118°C)
Cylinder head temp Green (0°C to 230°C)
RPM Set 1800 RPM
Suction Green (4.5 to 5.5 in Hg)
Oil pressure Green
Magneto drop Max 175 RPM, no more than 50 RPM difference between L and R
Propeller Check pitch control functional
Idling Check idling 800 RPM minimum

NOTE:

Oil Pressure During Normal Operation Because of the length of the oil flow between the sump and the oil pump there is a slightly lower oil pressure than would be expected. This effect will be more marked when the engine is cold and unlike a standard engine, the indicated oil pressure will normally tend to rise as the engine warms up. Thus it is not necessarily an indication of trouble if the oil pressure minima are only just met on start-up.



PILOTS NOTES FIREFLY T67M-MkII

3.5 PRE-TAKEOFF VITAL ACTIONS

Throttle friction	Stiff
Suction	Green (4.5 to 5.5 in Ha)
Oil temp/press	Green
Fuel press	Green
Pitot heater	On (if conditions require)
Horizon	Frect
DI	Synchronised = note wander
Strobe light	On
Magnetos	Roth
Fuel booster pump	On
Fuel contents	Check (Roth dauges)
Fuel cock	Check on /left or Pight Tank)
Flaps	Un or takeoff
	Check liftoff speed
	55 kts takeoff flap (18°)
	63 kts no flaps
Trim	Set at N
Harness	Tight and secure
Controls elev/ail	Full and free movement
Canopy latch	Closed position
	a.aaaa poblololi

PRE-TAKEOFF EMERGENCY BRIEF

The following points must be briefed:

- 1. Engine failure on the ground.
- 2. Engine failure below about 300 ft.
- 3. Engine failure above 300 ft.

The following points must be considered:

- 1. Runway surface type and condition.
- 2. Runway Tength.
- 3. Surface wind.
- 4. Availability of emergency landing areas round airfield.



TAKEOFF AND CLIMB

3.6	TAKEOFF AND CLIMB
Takeo	ff
	Throttle Full throttle
	RPM Check 2550 RPM minimum
	Oil pressure
	Oil temp
	Cylinder head temp
	ASI Increasing
	Raise nosewheel at 45 kts IAS
	Takeoff - liftoff speed 55 kts takeoff flap
	63 kts no flap
	Climb 70 kts takeoff flap (18°)
	77 kts no flap
	<u>WARNING</u>
>	IN STRONG CROSSWIND CONDITIONS LEAVE NOSEWHEEL ON THE GROUND UNTIL TAKEOFF SPEED THEN ROTATE TO TAKEOFF ATTITUDE.
After	Takeoff Checks
	Brakes On/off
	Flaps Raise at 73 kts
	Temps & press Check
	Booster pump Off (at a safe height)
	Fuel pressure Check
Depart	cure Checks

Altimeter Set as required

Temps & press Check



3.7 ERECT SPIN RECOVERY

3.7.1 Standard Recovery Technique

- a) Close the throttle.
- b) Raise the flaps.
- c) Check direction of spin on the turn co-ordinator.
- d) Apply full rudder to oppose the indicated direction of turn.
- e) Hold ailerons firmly neutral.
- f) Move control column progressively forward until spin stops.
- g) Centralise rudder.
- h) Level the wings with aileron.
- i) Recover from the dive.

WARNING

WITH C OF G AT REARWARD LIMIT THE PILOT MUST BE PREPARED TO MOVE CONTROL COLUMN FULLY FORWARD TO RECOVER FROM SPIN.

3.7.2 Incorrect Recovery

- A high rotation rate spin may occur if the correct recovery procedure is not followed, particularly if the control column is moved forward, partially or fully, BEFORE the application of full anti-spin rudder. Such out-of-sequence control actions will delay recovery, and increase the height loss. If the aircraft has not recovered within 2 complete rotations after application of full anti-spin rudder and fully forward control column, the following procedure may be used to expedite recovery.
 - a. Check that <u>FULL</u> anti-spin rudder is applied.
 - b. Move the control column $\underline{\text{FULLY AFT}}$ then $\underline{\text{SLOWLY FORWARD}}$ until the spin stops.
 - c. Centralise the controls and recover to level flight, (observing the "g" limitations).

3.7.3. Aerobatics or Spinning - Gyro Instruments

Aerobatics or spinning may cause the artificial horizon or directional gyro to topple. Up to $10\ \text{minutes}$ may be required for a gyro instrument to resume normal operation.



3.8 PRACTICE FORCED LANDINGS

Mixture rich.

Descend at 78 kts.

Warm engine and clear plugs every 1000 ft.

3.9 REJOIN CHECKS

Fuel contents Check (Both gauges Wing Tank A/C)

Fuel control Check on (Select tank with highest quantity Wing Tank A/C)

Engine Check gauges green. Mixture rich

DI Synchronise

Radio Select and check comms and navigation aids. Make joining call.

Altimeter Set correct millibar setting

3.10 LANDING CHECKS AND SPEEDS

3.10.1 Downwind Checks

Brakes Off - parking brake off

Engine Temps and press green. Mixture rich and locked. RPM to max

Fuel cock Check on (Left or Right Tank Wing Tank A/C)

Fuel contents Check (For tank selected Wing Tank

Booster pump On

Fuel pressure Check

Flaps As required

Altimeter QFE set

Harness Tight and locked



3.10.2 Circuit Speeds

	Normal and Glide		Flapless	
	Flap Position	Speed (Kts)	Flap Position	Speed (Kts)
Down Wind	Up	85	Up	85
Final Turn	Takeoff	75	Up	78
Finals	Landing	70	Up	
Threshold	Landing	70	Up	

3.10.3 Final Checks

Flap Set as required

Altimeter Correct QFE set

Landing Clearance received

.11 AFTER LANDING

.11.1 Checks After Landing

Landing light	0ff
Strobe light	0ff
Pitot heat	Off
Booster pump	0ff
Flaps	Uр

3.11.2 Stopping the Engine

RPM	.Increase to	1800 for	r 15–20 s	seconds,
; ¹	then reduce	to 1200	prior to	shut
	down.		•	
			•	
Throttle	Clased		•	





3.11.2 Stopping the Engine

	i e	
	Parking brake	. On
Run at 1000 RPM for I minute		
	Radios	. Off
	Navigation Aids	. Off
	Nav lights	. Off
	Alternator	Off (Check alternator fail warning operates)
	Magnetos	. Check for dead cut
	Throttle	. Closed
	Mixture	
	Magnetos	. When engine stops, Off
	Master switch	. Off
-	Fuel cock	. Off
	Flaps	Down
	Parking brake	Leave on if aircraft not chocked
3.11.3 Fitting Flying Control Lock (Mod 435)		
	Flaps	Select up
	Control lock	Fit to control sticks and flap operating lever (carefully move assembly into forward stick position)



INTENTIONALLY BLANK

P.3-14 CAA Approved July 1993 All TP.T67M-MkII/FM